

# Stress concentrations at an oblique hole in a thick plate

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## Summary

An empirical equation has been obtained for the elastic stress concentration factor at an isolated oblique circular- cylindrical hole in thick flat plate subjected to a uniform, arbitrarily oriented, uniaxial tension. The equation is presented and its development is outlined in this note.

**Keywords** Stress concentration factor; Empirical equation; Surface fit optimisation; Oblique hole; Thick plate.

## 1. Introduction

An oblique hole in a thick flat plate is shown in Fig.1. The hole is circular-cylindrical and the plate thickness is of the order of the hole diameter or greater; the applied stress is a uniform uniaxial tension in the plane of the plate. The plate edges are assumed to be remote from the hole. The definitive variables are

- (i) the angle  $\alpha$  between the hole-axis and the plate normal (i.e. the angle of obliquity);
- (ii) the ratio of plate thickness ( $h$ ) to hole diameter ( $d$ ),  $h/d$ ;
- (iii) the angle  $\phi$  between the direction of the applied stress and the major axis of the elliptic intersection of the hole in the surface plane of the plate.

The results of a parametric photoelastic study of the stress distribution at such a hole were presented in ref.1 for values of the variables  $h/d = 1.33, 2.00$  and  $3.00$  (approx.);  $\phi = 0, 30^\circ, 60^\circ$  and  $90^\circ$ ; and  $\alpha = 0, 15^\circ, 30^\circ, 45^\circ$ , and  $60^\circ$ . (For  $h/d = 2.00$ , the  $\phi = 30^\circ$  series was not included.)

Specific stress values were given in non-dimensional form as stress indices i.e. stress divided by the applied stress. The maximum stress index at each hole (i.e. the stress concentration factor), designated  $I_{\text{IMAX}}$ , is tabulated in ref. 1; values are reproduced here in Table 1 for ease of reference. Values for the case of  $h/d \approx 3.00$  are plotted in Fig.2a to illustrate typical trends.

Details of the positions of the maximum stress indices,  $I_{\text{IMAX}}$ , on the hole surface, are also given in ref. 1. For  $\alpha = 0$ , the maximum index occurs at the mid-plane position at the ends of the diameter normal to the applied load direction. With increasing obliquity,  $I_{\text{IMAX}}$  moves progressively from the mid-plane position, until for  $\alpha = 60^\circ$  the maximum occurs at the rim of the hole. In general, the position of  $I_{\text{IMAX}}$  is close to the point on the hole surface where the tangent is parallel to the applied load direction. However, for  $\alpha$  values greater than  $45^\circ$ , the position of  $I_{\text{IMAX}}$  is somewhat closer to the ‘acute’ end of the major axis of the elliptical intersection (end ‘a’ in Fig. 1) than the tangent position when  $\phi$  is not equal to  $90^\circ$ .

The development of an empirical relationship covering the Table 1 data is the subject of this note.

## 1.1 Notation

a, b	coefficients
d	hole diameter
h	plate thickness
$I_{\text{IMAX}}$	maximum experimental stress index
$I_{\alpha\phi}$	maximum predicted stress index
$I_{00}$	value of $I_{\alpha\phi}$ for $\alpha=\phi=0$
n	sample size

S.D	standard deviation of relative differences $\Delta$ about mean value $\bar{\Delta}$
$\alpha$	angle between hole axis and plate normal (see Fig.1)
$\Delta$	relative difference $(I_{\alpha\phi} - I_{1MAX})/I_{1MAX}$
$\bar{\Delta}$	mean value of $\Delta$ within a given data set
$\phi$	angle between direction of applied stress and major axis of the elliptical intersection of the hole in the surface plane of the plate (see Fig.1).

## 2. Assumed function

It was assumed that a satisfactory approximation to the experimental data could be obtained in the form of the quantity  $I_{\alpha\phi}$  derived from the following function

$$I_{\alpha\phi} = I_{00} + a \cos\phi (1 - \cos\alpha) + b \sin\phi (1 - \sec\alpha) \quad (1)$$

The  $I_{00}$  term is clearly to be identified with the common  $\alpha=0$  value of the curves in Fig. 2a; the second term (with negative a) is consistent with the monotonically decreasing curve for  $\phi=0$  and the third term (with negative b) with the monotonically increasing curve for  $\phi=90^\circ$ .

Another function, with the  $b \sin\phi (1 - \sec\alpha)$  term in equation (1) replaced by  $b \sin\phi \tan\alpha$ , was also studied, but in general the equation (1) function proved the more satisfactory.

## 3. Determination of $I_{00}$ , a and b

Values of the three fitting parameters ( $I_{00}$ , a and b) in equation (1) were determined for which the sum of the squares of the relative differences between the predicted values ( $I_{\alpha\phi}$ ) and the corresponding actual values ( $I_{1MAX}$ ) of the stress concentration factor in a particular data set ( $\sum [(I_{\alpha\phi} - I_{1MAX})/I_{1MAX}]^2$ ) was a minimum. For this purpose the experimental results and computations were tabulated using Microsoft Excel version 7.0 running on a Pentium PC. The

least squares fits were minimised using a Newton–Raphson search. The running time was negligible. The whole exercise entailed approximately 20 runs.

Two approaches were adopted. The first was to determine least-squares fit values of  $I_{00}$ ,  $a$  and  $b$  for each of the three separate data sets detailed in Table 1. For the  $h/d \approx 3.00$  set, as an example, the results were

$$I_{00} = 3.404$$

$$a = -3.047$$

$$b = -2.046$$

Values of  $I_{\alpha\phi}$  obtained from equation (1) using these values of  $I_{00}$ ,  $a$  and  $b$  are tabulated in Table 2a, and the relative differences  $[(I_{\alpha\phi} - I_{1MAX})/I_{1MAX}]$  between corresponding predicted and experimental values are given in Table 2b. It can be seen that these relative differences were within the range +6.1% to –6.0%. The mean value was –0.1% and the standard deviation about the mean (S.D.) was 3.6%.

Least squares fit values of  $I_{00}$ ,  $a$  and  $b$ , the relative difference range and the standard deviation of the relative differences are given in Table 3 for each data set.

It has been concluded previously [1] that

“For a given hole obliquity ( $\alpha$ ) and a given load direction ( $\phi$ ) the values of the principal stress indices ( $I_{1MAX} \dots$ ) are not significantly dependent upon the ratio of plate thickness ( $h$ ) to hole diameter ( $d$ ) over the range  $1.33 < h/d < 3.0$ ”

With this in mind, as a second approach, least squares fit values of  $I_{00}$ ,  $a$  and  $b$  were obtained for the average  $I_{MAX}$  values for each  $\alpha, \phi$  combinations in Table 1, and then, using these  $I_{00}$ ,  $a$  and  $b$  values, the relative difference distribution, mean and standard deviation for each separate data set were obtained.

Least squares fit values for the average data set were:

$$I_{00} = 3.359$$

$$a = -2.884$$

$$b = -2.238$$

Average experimental values, values obtained from equation (1) using the above values of  $I_{00}$ ,  $a$  and  $b$ , and relative differences are tabulated in Table 4. The relative difference range was +6.0% to -6.3%; the mean value was -0.34%, and the standard deviation was 3.7%. The relative difference range, mean and standard deviation for each separate data set, obtained using the same values of  $I_{00}$ ,  $a$  and  $b$ , are given in Table 5. It can be seen that the equation

$$I_{\alpha\phi} = 3.36 - 2.88\cos\phi (1-\cos\alpha) - 2.24\sin\phi (1-\sec\alpha) \quad (2)$$

provides maximum stress indices which differ from experimental values by no more than +9.0% to -7.8%. The greatest mean relative difference value was -1.6% and the standard deviations were approximately 4%.

To facilitate comparison with experimental data, the variations of  $I_{\alpha\phi}$  with  $\alpha$  for  $\phi=0, 30^\circ, 60^\circ$  and  $90^\circ$ , as derived from equation (2), are shown in Fig. 2b.

#### 4. Discussion

There are 47 independent data items in Table 1. The distribution of the relative differences between values of  $I_{\alpha\phi}$  calculated from equation (2) and corresponding values of  $I_{I\text{MAX}}$  is shown in Fig 3 (the plotted 'number' is the number of relative differences greater than the indicated percentage value). The average relative difference is  $-0.5\%$ . 36 of the 47 relative differences (77%) are less than 5% (numerically) and only 6 (13%) are greater than 7%. The combination of  $\alpha$  and  $\phi$  at which the extreme difference values tabulated in Table 5 occurred appeared to be randomly distributed; there was no indication of a tendency for the extreme values to occur at a particular co-ordinate combination.

It follows from the form of equation (1) that the gradient of the  $I_{\alpha\phi}$  versus  $\alpha$  curve is zero at  $\alpha=0$  for all  $\phi$ . It can be shown that for all  $\phi$  values less than  $\tan^{-1}a/b$ , a second zero gradient (i.e. a turning point or minimum) will also occur in the curve. As a consequence, using the  $a$  and  $b$  values in equation (2), such minima are predicted in the  $\phi=0$  and  $\phi=30^\circ$  curves. For  $\phi=0$  this minimum occurs at  $\alpha=90^\circ$ , but since no physical significance can be ascribed to such a hole the deduction is meaningless. For  $\phi=30^\circ$  the minimum is predicted to occur at  $\alpha=48^\circ$ . A minimum in the  $I_{I\text{MAX}}$  curve for  $\phi=30^\circ$  can be seen in Fig.2a in the region of  $\alpha=45^\circ$ . (A similar minimum occurred in the  $h/d=1.33$  data [1].) The capability of equation (2) to predict this feature of the experimental data is noteworthy. The minimum  $I_{\alpha\phi}$  value for  $\phi=30^\circ$ ,  $\alpha=48^\circ$  was approximately 6% greater than the value indicated in Fig. 2a.

## 5. Conclusion

Within the range of variables  $1.3 < h/d < 3.00$  and  $\alpha \leq 60^\circ$ , the equation

$$I_{\alpha\phi} = 3.36 - 2.88\cos\phi (1 - \cos\alpha) - 2.24\sin\phi (1 - \sec\alpha)$$

yields estimates of the maximum stress index (i.e. stress concentration factor) at an oblique hole in a flat plate subjected to a uniform uniaxial tension which differ by no more than 9% from experimental values.

## Reference

1. Stanley, P. and Day, B.V., “ Photoelastic investigation of stresses at an oblique hole in a thick flat plate under uniform uniaxial tension”, J. Strain Anal., 1990,25,157-175

**Table 1:** Experimental data ( $I_{MAX}$ )

$\phi$ (deg)	h/d	$\alpha$ (deg)				
		0	15	30	45	60
0	1.33	3.31	3.16	2.87	2.52	2.08
	2.00	3.26	3.15	3.12	2.44	1.93
	3.00	3.38	3.20	2.94	2.40 <sup>1</sup>	1.99 <sup>2</sup>
30	1.33	—	3.23	2.98	2.95	3.30
	2.00	—	—	—	—	—
	3.00	—	3.25	3.26	2.88 <sup>1</sup>	3.12 <sup>2</sup>
60	1.33	—	3.38	3.60	3.65	4.78 <sup>3</sup>
	2.00	—	3.56	3.55	3.71	4.75
	3.00	—	3.47	3.68	3.64 <sup>4</sup>	4.20 <sup>5</sup>
90	1.33	—	3.73	3.96	4.32	5.64 <sup>3</sup>
	2.00	—	3.73	3.88	4.37	5.61
	3.00	—	3.55	3.90	4.52 <sup>4</sup>	5.63 <sup>5</sup>

Notes

1: h/d = 2.80  
2: h/d = 2.59  
3: h/d = 1.20  
4: h/d = 2.90  
5: h/d = 2.57

**Table 2a:**  $I_{\alpha\phi}$  from equation (1) for h/d  $\approx$  3.00

$\phi$ (deg)	$\alpha$ (deg)				
	0	15	30	45	60
0	3.40	3.30	3.00	2.51	1.88
30	3.40	3.35	3.21	3.05	3.11
60	3.40	3.41	3.47	3.69	4.41
90	3.40	3.48	3.72	4.25	5.45

**Table 2b:**  $(I_{\alpha\phi} - I_{MAX})/I_{MAX} \times 100$  for h/d  $\approx$  3.00

$\phi$ (deg)	$\alpha$ (deg)				
	0	15	30	45	60
0	0.7%	3.1%	1.9%	4.6%	-5.5%
30	0.7%	3.1%	-1.6%	6.1%	-0.4%
60	0.7%	-1.6%	-5.6%	1.4%	5.1%
90	0.7%	-2.1%	-4.6%	-6.0%	-3.2%



**Table 3:** Least squares fit values of  $I_{00}$ , a and b, relative difference ranges, mean values and standard deviations

h/d	$I_{00}$	a	b	Difference range	Mean	S.D.
1.33	3.320	-2.667	-2.323	+7.1% -8.8%	-0.14%	3.88%
2.00	3.361	-2.888	-2.374	+3.6% -7.6%	-0.12%	3.61%
3.00	3.404	-3.047	-2.046	+6.1% - 6.0%	-0.12%	3.62%

**Table 4:** a) averaged experimental data  
b) calculated  $I_{\alpha\phi}$  data  
c) relative differences (%)

(a)	$\phi$ (deg)	$\alpha$ (deg)				
		0	15	30	45	60
	0	3.317	3.170	2.977	2.453	2.000
	30	3.317	3.240	3.120	2.915	3.210
	60	3.317	3.470	3.610	3.667	4.577
	90	3.317	3.670	3.913	4.403	5.627
(b)	$\phi$ (deg)	$\alpha$ (deg)				
		0	15	30	45	60
	0	3.36	3.26	2.97	2.51	1.92
	30	3.36	3.31	3.20	3.09	3.23
	60	3.36	3.38	3.47	3.74	4.58
	90	3.36	3.44	3.71	4.29	5.60
(c)	$\phi$ (deg)	$\alpha$ (deg)				
		0	15	30	45	60
	0	1.3%	2.9%	-0.1%	2.5%	-4.1%
	30	1.3%	2.3%	2.5%	6.0%	0.6%
	60	1.3%	-2.6%	-4.0%	2.0%	0.0%
	90	1.3%	-6.3%	-5.3%	-2.7%	-0.5%

**Table 5:** Least squares fit values of  $I_{00}$ , a and b, from averaged data, with relative difference ranges, means, and standard deviations

h/d	$I_{00}$	a	b	Difference range	Mean	S.D.
average	3.360	-2.884	-2.238	+6.0% -6.3%	-0.34%	3.37%
1.33	“	“	“	+7.3% -7.8%	-0.51%	4.41%
2.00	“	“	“	+3.5% -7.8%	-1.59%	3.56%
3.00	“	“	“	+9.0% - 5.8%	+0.21%	4.36%

## Figure captions

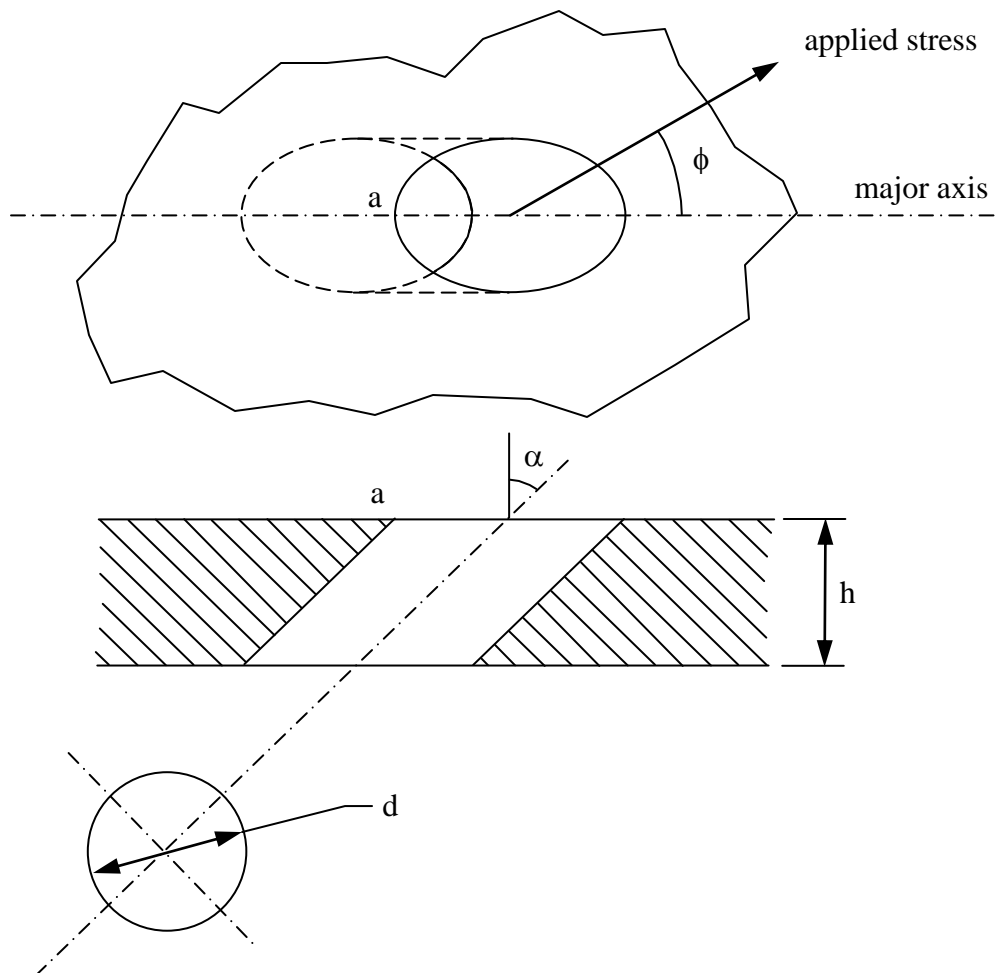
**Figure 1:** Oblique hole in plate subjected to uniaxial stress

**Figure 2a:** Experimental data for  $h/d \approx 3.00$

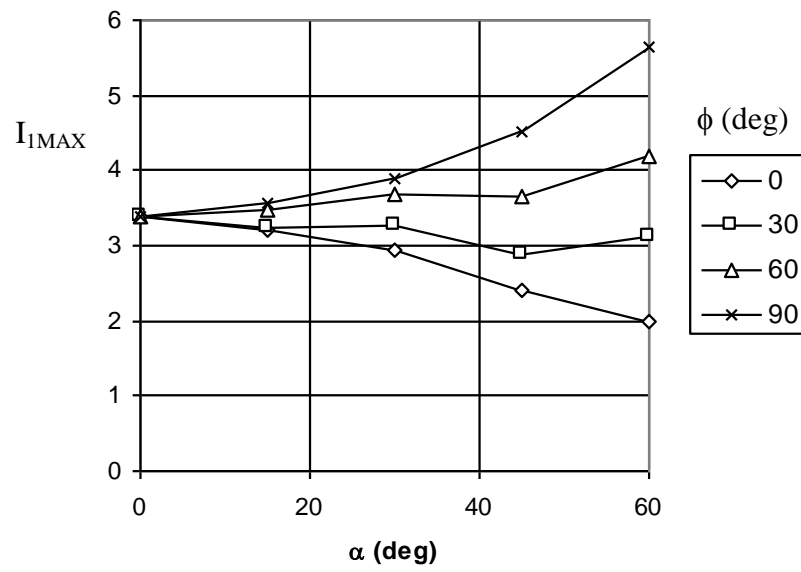
**Figure 2b:** Maximum stress indices derived from equation (2).

**Figure 3:** Number of  $\Delta$  values numerically greater than indicated percentage value  
(Note: The values of 20 and 27 at 0% are the numbers of positive and negative relative differences, respectively.)

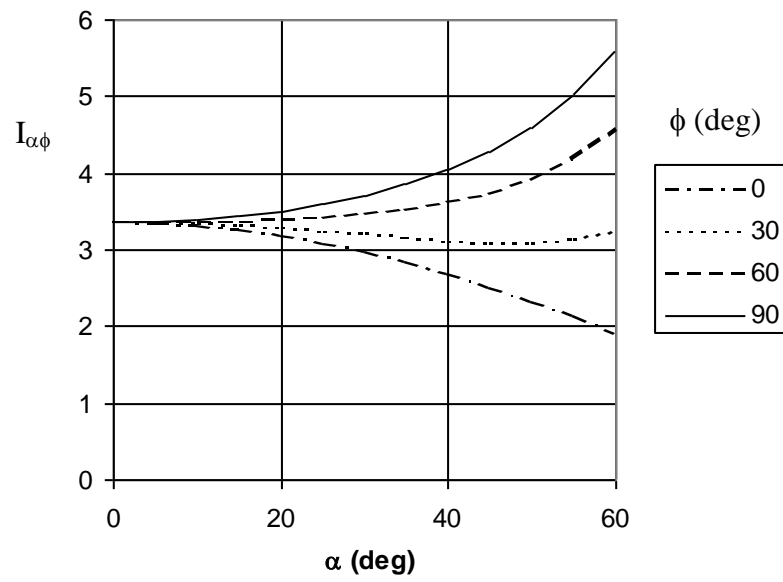
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